THERMAL ENTRANCE REGION UNDER HIGHLY NON-ISOVISCOUS FLOW*

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Received October 26th, 1972

The problem has been solved of simultaneous heat and momentum transfer in an entrance region of a heat exchanger for liquids with highly temperature-dependent viscosity. The method of thermal boundary layer has been applied using a linear approximation of the temperature and a velocity profile corresponding precisely to the temperature. A solution to the boundary layer equation has been outlined and explicit expressions introduced for heat transfer and pressure drop valid for short exchangers with a thin boundary layer.

Heat transfer under forced convection into a liquid flowing in a tube of different wall temperature may be usually described by the equation

$$\rho c_p v_x \frac{\partial T}{\partial x} = k \left[\frac{\partial^2 T}{\partial r^2} + \frac{1}{r} \frac{\partial T}{\partial r} \right], \tag{1}$$

with the boundary conditions: $T = T_0$ for x = 0, and $T = T_W$ for r = R and x > 0. (2), (3)

Even in cases when v_x depends both on r and the axial coordinate x one may often neglect the effect of the radial velocity component and accept Eq. (1). Solution of this equation mandates some numerical method, mostly that of finite differences.

For short exchangers (where $x < 0.1 R^2 \rho c_p U/k$) it is convenient to utilize the method of thermal boundary layer based on assumption that the resistance to heat transfer occurs only in a thin layer adherring to the wall. The set of Eqs (I) - (3) may then be supplemented by another condition, namely

$$T = T_0$$
 and $\partial T / \partial r = 0$ for $r = R - \delta$, (4)

permitting a simplified solution to Eq. (1); the function $\delta(x)$ is calculated afterwards. The results obtained by exact solution of the thermal boundary layer equation (provided that the temperature profile in the boundary layer is given by the first eigenfunction of an equation obtained from Eq. (1) (ref.^{1,2}) by separation) are in an excellent agreement with the numerical results of the complete set of Eqs (1)–(3) both for isoviscous³ and non-isoviscous⁴ flow. The given approach however, requires numerical approach even to the boundary layer although substantially simpler^{1,2} than that for the complete set. A principal advantage of the boundary layer method should be the possibility of obtaining by simple mathematical means an approximate, yet reliable solution.

Part V of the series Heat Transfer in Laminar Flow; Part IV: This Journal 38, 3094 (1973).

Such a solution of non-isoviscous heat transfer by the boundary layer method has been attempted by Murakami⁵ who used a very close approximation of the temperature profile (third order power expression) but his superimposed velocity profile resulted in obviously unsound conclusions. The submitted paper though utilizing a rough assumption about the temperature profile (linear profile within the boundary layer) uses a velocity profile that corresponds precisely to the viscosity and shear stress distribution. According to our experience with isoviscous heat transfer this assumption provides a reliable fit to the conditions in short exchangers; the systematic error amounts to about 10%. The proposed method enables to solve simultaneously the problem of coupled heat and momentum transfer analytically with an explicit expression of the results.

A Solution of the Velocity and Temperature Profile in the Boundary Layer

The course of the shear stress across the tube is given by a linear expression as

$$\tau = \tau_{s} r / R$$
, (5)

where τ_s is the shear stress on the wall

$$\tau_s = \frac{1}{2} (dp/dx) R . \qquad (6)$$

The velocity gradient in a Newtonian liquid is

$$dv_x/dr = -\tau/\mu(r).$$
(7)

In accord with the concept of the thermal boundary layer of thickness δ with the temperature varying linearly between T_w and T_o

$$T = T_w + (T_0 - T_w)(R - r)/\delta \quad \text{for} \quad r \ge R - \delta \tag{8}$$

$$T = T_0 \quad \text{for} \quad r \leq R - \delta \,, \tag{9}$$

the course of viscosity in the boundary layer is given by

$$\mu = \mu_{\rm W} \exp\left[\psi(R-r)/\delta\right]. \tag{10}$$

A criterion of isoviscousness, ψ , is determined from the temperature dependence of the viscosity expressed in the investigated range with the aid of an empirical formula

$$\mu = \mu_0 \exp\left(-AT\right). \tag{11}$$

The dimensionless variables are introduced by

$$w = v_x/u, \quad b = (dp/dx) R^2/(8\mu_w U), \quad y = (R - r)/R, \quad y_0 = \delta/R.$$
(12)-(15)

Collection Czechoslov. Chem. Commun. [Vol. 39] [1974]

Combining the definitions (12)-(15) with Eqs (5)-(10) one obtains two differential equations for the velocity profile. For their solution one uses the "no-slip" condition on the wall

$$w = 0$$
 for $y = 0$ (16)

and the condition of constant flow rate

$$\int_{0}^{1} w(1 - y) \, \mathrm{d}y = \frac{1}{2} \,. \tag{17}$$

On defining the auxiliary variables u_1 and u_2 as

$$u_1(y) = w(y)/(4b)$$
 for $y \le y_0$, $u_2(y) = w(y)/(4b)$ for $y \ge y_0$. (18), (19)

the solution of the differential equations may be expressed in terms of these variables as

$$u_1(y) = \int_0^y (1-y) \exp(\psi y/y_0) \, \mathrm{d}y \,, \quad u_2(y) = u_1(y_0) + \int_{y_0}^y (1-y) \exp(-\psi) \, \mathrm{d}y \,.$$
(20), (21)

Denoting further by I1, I2, I3 the following definite integrals

$$I_{1} = \int_{0}^{y_{0}} u_{1}(y) y/y_{0} (1 - y) dy, \qquad (22)$$

$$I_{2} = \int_{y_{0}}^{1} u_{2}(y) (1 - y) dy, \qquad (23)$$

$$I_{3} = \int_{0}^{y_{0}} u_{1}(y) (1 - y) dy. \qquad (24)$$

the velocity, pressure drop and cup-mixing temperature can be expressed as

$$w(y) = u_1(y)/(I_2 + I_3)$$
 for $y \le y_0$, (25)

$$w(y) = u_2(y)/(I_2 + I_3)$$
 for $y \ge y_0$, (26)

$$b = 1/[8(I_2 + I_3)], \qquad (27)$$

$$t_{\rm M} = ({\rm I}_1 + {\rm I}_2)/({\rm I}_2 + {\rm I}_3).$$
 (28)

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In the calculation of the thickness of the thermal boundary layer, y_0 , as a function of y we use the integral balance on heat, which for the given temperature gradient in the boundary layer may be written in dimensionless form as

$$dt_{M}/dz = -2/y_{0}, \qquad (29)$$

from which

$$z(y_0) = \frac{1}{2} \int_0^{y_0} y_0(- dt_M/dy_0) dy_0 .$$
 (30)

Thus it suffices to differentiate the function $t_{M}(y_0)$ determined from Eq. (29) and the result is obtained by inversion of the function calculated by integrating Eq. (30).

Asymptotic Solution for Low z

Each integration in Eqs (20)-(24) may be carried out analytically. Moreover, if one neglects in the limit $y_0 \rightarrow 0$ higher powers of y_0 the expressions in Eqs (25)-(28)may be given by particularly simple functions of y_0 and ψ . Thus then dimensionless cup-mixing temperature is

$$t_{\rm M} = 1 - \frac{4}{3} y_0^2 \left[1 - f(\psi) \right], \qquad (31)$$

where

$$f(\psi) = 1 + 6 \left[\frac{1}{\psi^3} - \exp\psi\left(\frac{1}{\psi^3} + \frac{1}{2\psi} - \frac{1}{\psi^2}\right) \right].$$
 (32)

On substituting Eq. (31) into Eq. (30), integrating and inverting we get

$$y_0 = \left[\frac{9z}{4[1 - f(\psi)]}\right]^{1/3}.$$
 (33)

The last equation permits the variables calculated for $y_0 \rightarrow 0$ from Eqs (27) and (28) to be expressed as functions of z:

$$t_{\rm M} = 1 - \frac{4}{3} (9/4)^{2/3} z^{2/3} [1 - f(\psi)]^{1/3} , \qquad (34)$$

$$b = \exp \psi \{ 1 - z^{1/3} 4(9/4)^{1/3} [1 + (1/\psi) - (\exp \psi/\psi)] / [1 - f(\psi)]^{1/3} \}^{-1}.$$
 (35)

An interesting result is the value of the coefficient α defined as

d ln (Nu_{noniso}/Nu_{iso})/d ln
$$\left(\frac{\mu_{\rm w}}{\mu_{\rm M}}\right) = -\ln\left[1 - f(\psi)\right]/(3\psi)$$
. (36)

Collection Czechoslov. Chem. Commun. [Vol. 39] (1974)

The course of this coefficient is plotted in Fig. 3. As it may be seen α does not depend strongly on ψ and hence one can accept the following relation

$$Nu_{noniso} = Nu_{iso}(\mu_W/\mu_M)^{0.25}, \qquad (37)$$

where 0.25 is the value of α at $\psi \rightarrow 0$. The high value of the exponent (0.25) instead





Dimensionless Cup-Mixing Temperature under Non-Isoviscous Flow in Short Exchangers





Dimensionless Local Pressure Drop as a Function of z

Solution by boundary layer method^{1,2} — — — approximative solution for low z (Eq. (35)), — . — . — . — pressure drop corresponding to inlet viscosity. Figures on curves indicate values of ψ .



FIG. 3 Calculated Exponent of the Sieder-Tate correction as a Function of ψ for $z \rightarrow 0$.

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of the more usual $\alpha = 0.14$ appearing in the correlation of the Sieder-Tate type⁶ (Eq. (37)) will be suitable just for the exchangers with short time of contact (small z). This is *e.g.* the case of mixed vessels with laminar regime near the wall (this occurs even for fairly high values of the Reynolds number for mixing), where, as mentioned by Uhl⁷, the experimentally found values of the exponent were $\alpha = 0.24$ and $\alpha = 0.25$. The often used value $\alpha = 0.14$ stems from experiments of Sieder and Tate for long tubular exchanger⁶.

DISCUSSION

1) The validity of the above solution may be judged for instance from comparison of Eq. (34) for isoviscous conditions:

$$t_{\rm M} = 1 - 2 \cdot 29 z^{2/3} \tag{34a}$$

with the exact solution due to Levéque8

$$t_{\rm M} = 1 - 2.56z^{2/3} \tag{38}$$

The relative error dt_M/dz amounts to 10.5%.

2) The agreement with the solution of non-isoviscous heat transfer may be judged from Figs 1 and 2 plotting the results obtained by solving the boundary layer equation under more realistic assumptions^{1,2}. As it is apparent the results of the complete set Eqs (1)-(3) and $(5)-(7)^{4,9}$ obtained by a numerical technique comport well with those of the presented method for higher values of z. At low values of z (where y_0 becomes comparable with the increment of the numerical computational grid) the numerical results lose their accuracy, while Eqs (31)-(37) hold in this very region.

3) The whole procedure can be repeated for non-Newtonian liquids, whose non-isoviscous flow is governed by

$$dv_{x}/dr = -(\tau/\eta_{0})^{1/n} \exp(AT), \qquad (39)$$

The procedure becomes particularly simple if 1/n is an integer¹⁰. The result is

$$t_{\rm M} = 1 - 2 \cdot 29 \left(\frac{3n+1}{4n}\right)^{1/3} z^{2/3} \left[1 - f(\psi)\right]^{1/3}, \tag{40}$$

where the correction of non-Newtonian behaviour is identical to that derived by Pigford¹¹. Furthermore, this case confirms theoretically the appropriateness of the

definition of the parameter A for non-Newtonian liquids as

$$A = \frac{\partial \ln \eta}{\partial t} \bigg|_{t = \text{const.}}$$
(41)

since it enables the effects of non-Newtonian behaviour and that of non-isoviscousness on heat transfer to be separated.

4) The above presented method is suitable, as it is apparent from Figs 1 and 2, for exchangers satisfying $z \ll 0.001$. The other stipulation is that $z \gg Pe^{-3/2}$ for we could neglect the effect of axial conduction on the over-all heat transfer¹². Under low values of Pe (*e.g.* for slow motion of liquid metals in capillaries) even the boundary conditions in Eqs (2) and (3) are difficult to realize. From the continuity equation and the velocity profiles (25) it further follows that for exchangers having $z \ll |\psi| Pe^{-3/2}$ one has to take into consideration also the effects of radial velocity and inertia forces.

Nevertheless, the range $Pe^{-3/2} \ll z \ll 0.001$ into which the presented results are confined covers a broad class of exchangers of practical interest; as has been shown some of the results hold not only for flow in tubes but also for agitated vessels used for heat transfer operations.

LIST OF SYMBOLS

A	material constant, Eq. (11), (41), (deg ⁻¹)	
b	dimensionless pressure drop	
c _n	heat capacity, $(cal g^{-1} deg^{-1})$	
f	function defined by Eq. (32)	
I_1, I_2, I_3	integrals defined by Eqs (22)-(24)	
k	thermal conductivity, $(cal cm^{-1} s^{-1} deg^{-1})$	
n	flow index	
Nunoniso	average Nusselt number based on arithmetic average temperature difference	
Nuise	Nusselt number for isoviscous flow	
dp/dx	local pressure drop, $(g cm^{-3} s^{-2})$	
$Pe = 2c_n UR/k$ Peclet number		
r	radial coordinate, (cm)	
R	radius of tube, (cm)	
Т	temperature, (deg C)	
T_0	inlet temperature of liquid, (deg C)	
$T_{\mathbf{W}}$	wall temperature, (deg C)	
$t_{\rm M} = (T_{\rm M} - T_{\rm W})/(T_{\rm 0} - T_{\rm W})$ dimensionlesss cup-mixing temperature		
T _M	cup-mixing temperature, (deg C)	
U	average velocity, (cm s ⁻¹)	
<i>u</i> ₁ , <i>u</i> ₂	auxiliary functions defines by Eqs. (18), (19)	
v,	axial component of velocity, (cm s ⁻¹)	
141	dimensionless velocity	

x	axial coordinate of examined cross section, (cm)
у	dimensionless distance from wall
y _o	dimensionless thickness of boundary layer
z = 2x/(1	R Pe) dimensionless axial coordinate
α	exponent of Sider-Tate correction, Eq. (36)
δ	thickness of thermal boundary layer, (cm)
η	apparent viscosity of non-Newtonian liquid (g cm ⁻¹ s ⁻¹
η_0	material constant, (g cm ^{-1} s ^{2$-n$})
μ	viscosity, $(g \text{ cm}^{-1} \text{ s}^{-1})$
μ_0	material constant, $(g cm^{-1} s^{-1})$
μ _M	viscosity at temperature $T_{\rm M}$, (g cm ⁻¹ s ⁻¹)
μ_{W}	viscosity at temperature T_{W} , (g cm ⁻¹ s ⁻¹)
0	density, $(g \text{ cm}^{-3})$
τ	shear stress, $(g \text{ cm}^{-1} \text{ s}^{-2})$
τw	wall shear stress, $(g \text{ cm}^{-1} \text{ s}^{-})$

 $\psi = A(T_w - T_0)$ criterion of non-isoviscousness

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Translated by V. Staněk.